Nascap-2k Spacecraft Surface Charging Code (EAR-controlled, freely available to US citizens and companies)

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100 YEARS OF U.S. AIR FORCE SCIENCE & TECHNOLOGY

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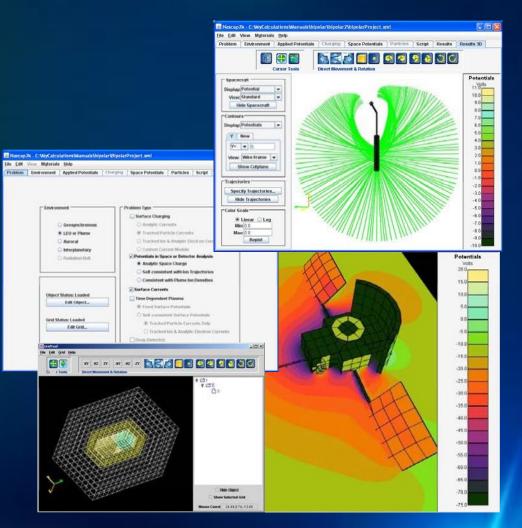
What Is Nascap-2k?

- *Nascap-2k* is a full 3-D fully GUI computer code to calculate the interactions of a spacecraft with its LEO, GEO, Polar, or interplanetary plasma environment. Interactions include:
 - Surface potentials
 - Charged particle spectra (electrons and ions)
 - Secondary electron emission
 - Photoemission
 - Bulk and surface conductivity
 - Current collection
 - Applied frame and distributed potentials
 - Perturbation of near-field environment
 - Space Potentials
 - Thruster plumes
 - Charge exchange ion flow
- Calculations are analytic where possible, PIC (particle-in-cell) where desired
- Code features
 - Object and grid definition
 - Simplified user interface for problem definition and execution
 - Graphical display of results
- Air Force Research Laboratory and NASA sponsorship
- Resides on your own computer, no web-login necessary



Nascap-2k Core Capabilities

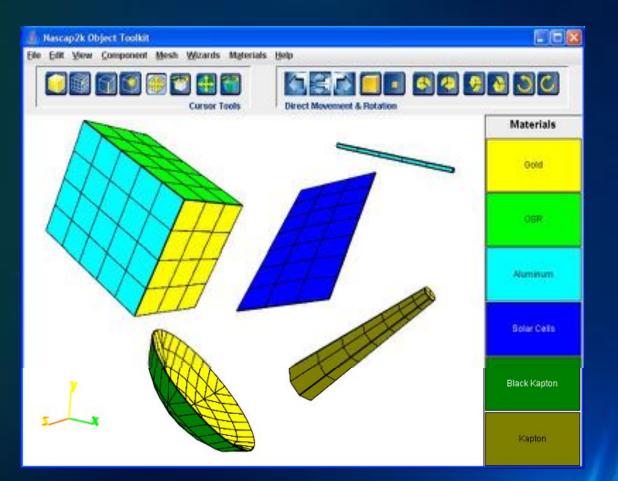
- Defines spacecraft surface geometry
- Grids space surrounding spacecraft
- Calculates environmentally induced time-dependent surface potentials
- Calculates external potentials:
 - Analytic space charge (5 models)
 - Macroparticle space charge (4 models)
- Generates and tracks macroparticles
 - Uniform with boundary injection
 - Sheath generation
 - Charge exchange
- Post-processing:
 - Time-dependent surface potentials and currents
 - Time-dependent volume potentials, currents, and densities





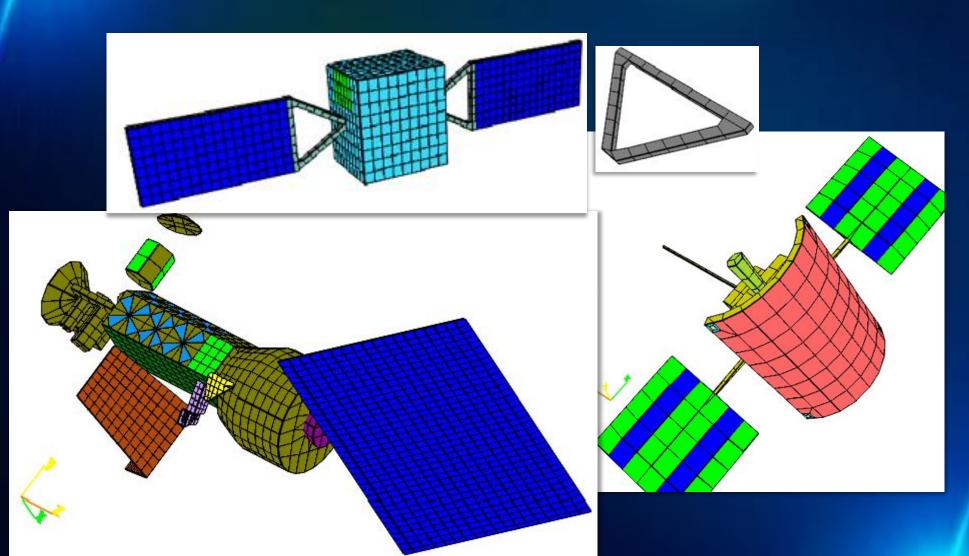
Object Toolkit

- Builds spacecraft surface
 models
- Uses intrinsic building blocks
- Imports from finite-element preprocessors
- Surface attributes from database or user-definable
 - Material name
 - Conductivity
 - Secondary electron emission
 - Thickness, etc.
 - Conductor number
- Customizable to other applications via an external file





Object Toolkit Examples

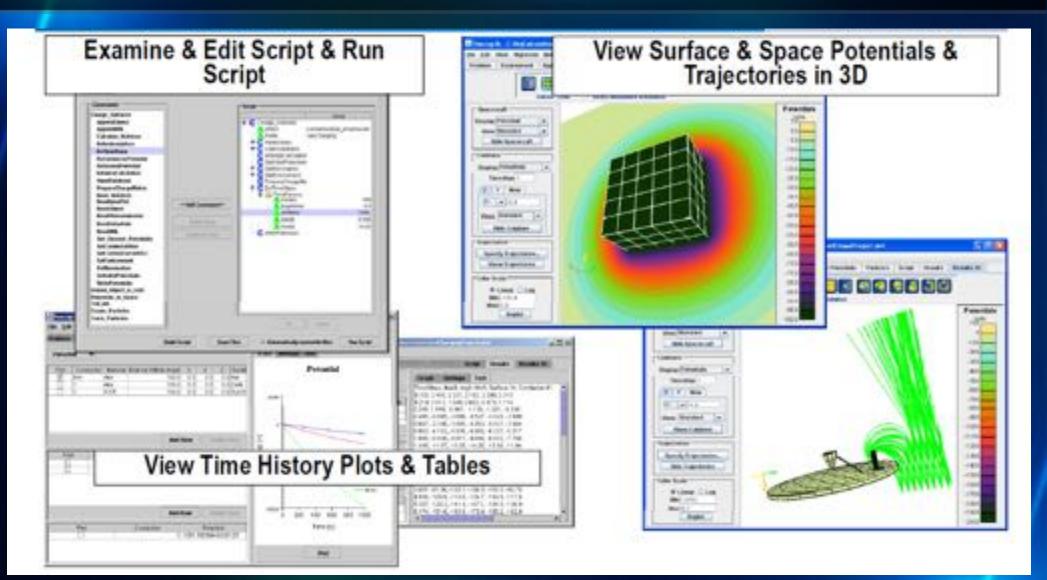




Nascap-2k GUI Input tabs

Nascap2k - D:MyCalculations/Manuals/bipolar/ Elle Edit View Materials Help Problem Environment Applied Potentials	Problem			
Environment Environment Coopercharpson LEO or Phane Aurorat Iter planetary Praductor Dett Object Status: Loaded Edit Object. Grid Status: Loaded Edit Grid.	Problem Type Surface Charging Anadylic Col Rescap2k - E:WEXTXTwoNSTARs\TwinThrust	LEO Environment LEO Environment Particle Problem Environment Particle Problem Environment Charge Density Model Laplace Problem Charge Density Model Laplace Problem Frozen ton Sencer Problem Charge Density Model Frozen ton Sencer Fr	Charge Density Formulation	Particle Initialization and Tracking
			Tracking	time per timestep: [1 000E-6]

Nascap-2k Results Tabs

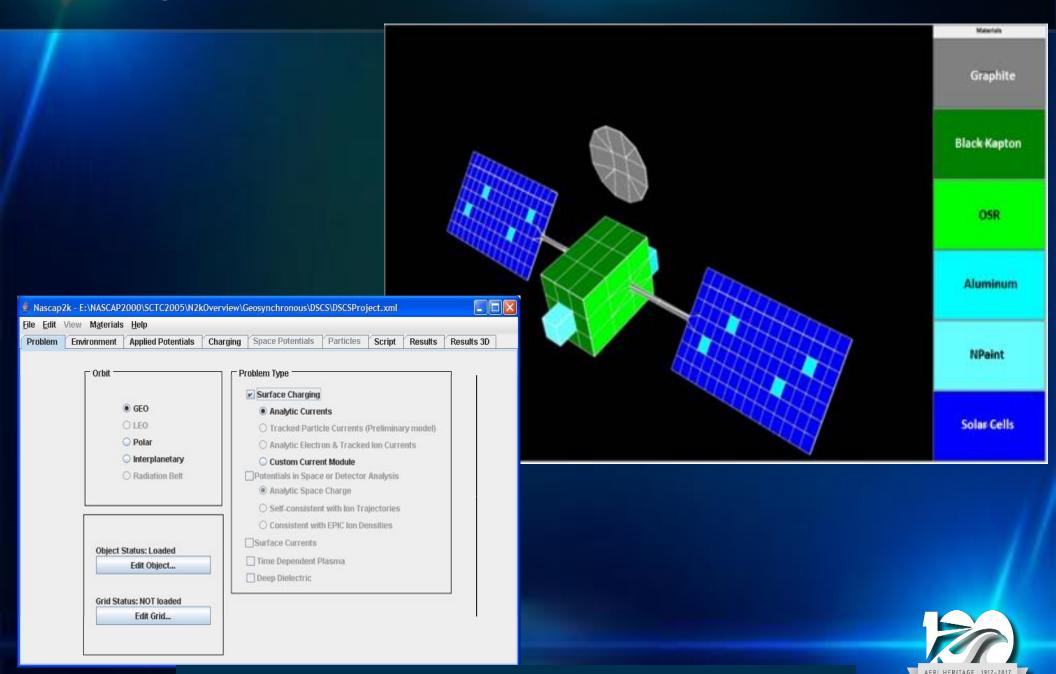




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Example 1 – GEO Spacecraft Model – 3-axis stabilized Defining the Problem and Surface Materials



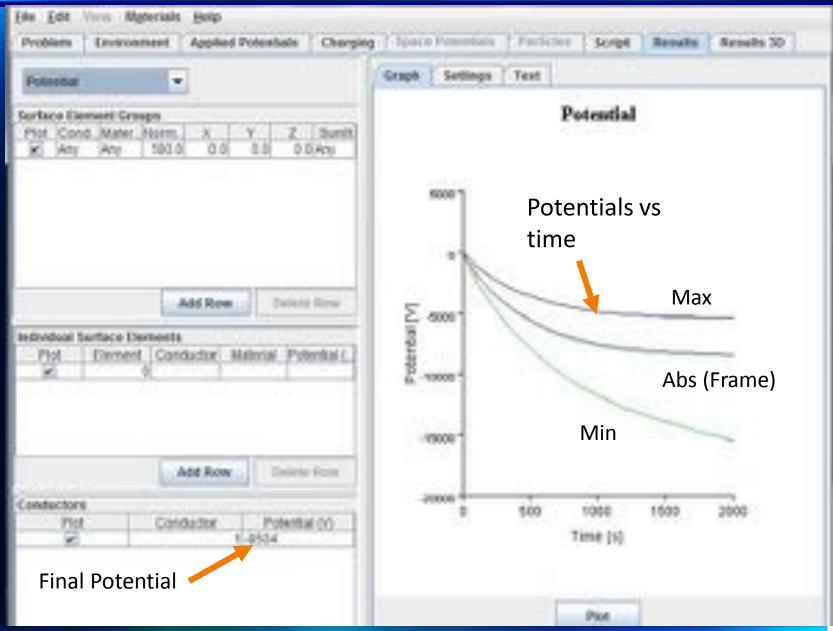
Example - Environment Input for Nascap-2k



trees Materials Help 644 Environment Applied Potentials Charging Space Potentials Permiting Script Results Results 30 Problem. Sun or shade May be Geosynchronous Environment single or GED Environment Plasma double r* Direction to Sen 875.6 w Jie to X-0.707 ¥10-0 Maxwellian Relative* Sun Intennity(1000 or Kappa Electron Density (m⁻¹): 1.220E1 Yorke at Spaceword's (Joshue at Earth Orbit) 22 distribution The philippe sector spectra Electron Temperature JeVy 1 600E4 1030 ion Donaity (m¹⁶) 2 34565 3175 Particle Species Tiple Mass (arrup) Charge (C) ion Temperature JeV5/2.050E4 1111.0 100.0 FIELDOR 108-8 Unknown 1.0026-10 Electron Current (Am⁹z-4 1376-6 Ion Current (Am²):2542E-8 lon species Magnetic Field (T) Bitte F fixio. Halo. Add Species Debete Species B field

Example - Charging History from Nascap-2k



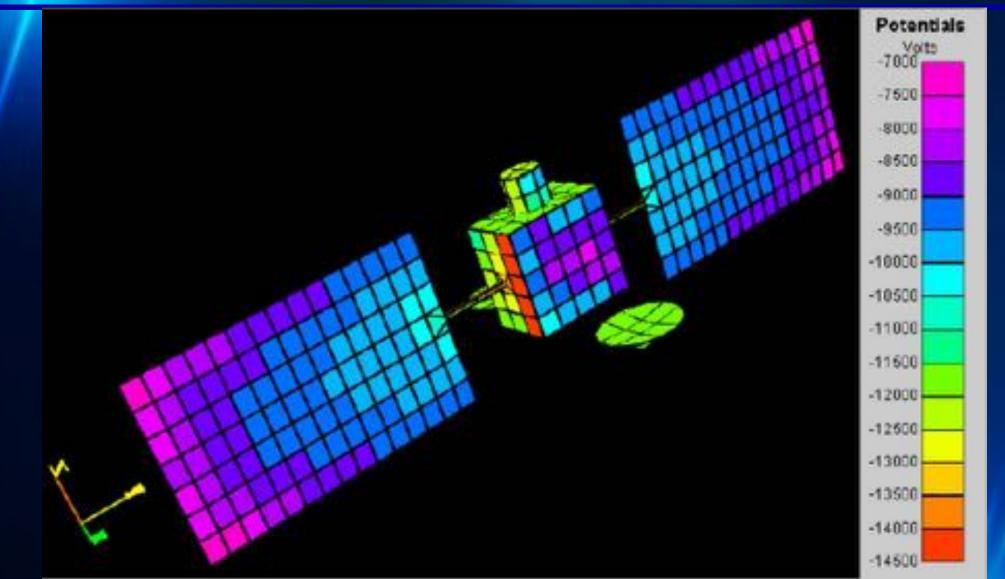


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Example - Surface Charging from Nascap-2k



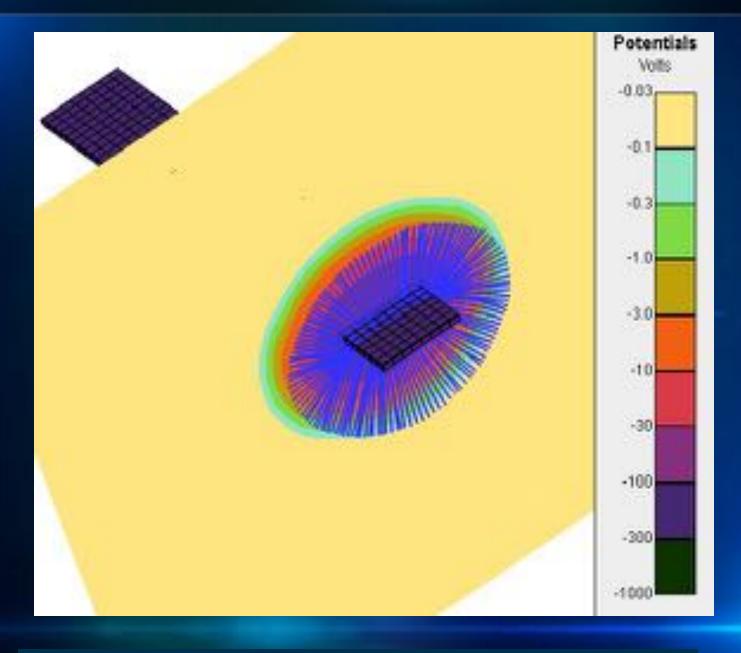




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3-D Results Display Surface Potentials, Space Potentials, and Ion Trajectories



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Geosynchronous Orbit Charging

Environment and Timestepping Specifications



Environment Applied Potentials Charging Sp	ce Potentials Particles Script Results Results 30	
GEO	Environment	
eo Environment Plasma	Sun	
Worst Case	Direction to Sun	
THE STREET	X: 1000 Y: 00 Z: 0.0	
htseveriliaiv	Relative* Sun Intensity: 1 000	
Electron Density (m ⁻³): 1.120E6	"(value at Spacecraft) / (value at Earth Orbit)	
lectron Temperature (eV): 1.200E4	Use photomucsico spoctra	
	· · · · · · · · · · · · · · · · · · ·	
Ion Density (m ⁻³); 2.360E5		
Ion Temperature (eV): 2.950E4		
Electron Current (Am ²): 3.289E-6		
Ion Current (Am ²);2.5368-8	Charging Time	
The second second second second	Start Time (sec)	End Time (sec):2000
ngnetic Field (T)	Min Timestep (sec)	Max Timestep (sec):50
Bx:0.0 By:0.0 Bz:0.0	Number of Timesteps*:45	
	"(Use only for Analytic Currents)	

- Environment tab sets geosynchronous environment distribution function-here a Maxwellian
- Charging tab sets timestep parameters

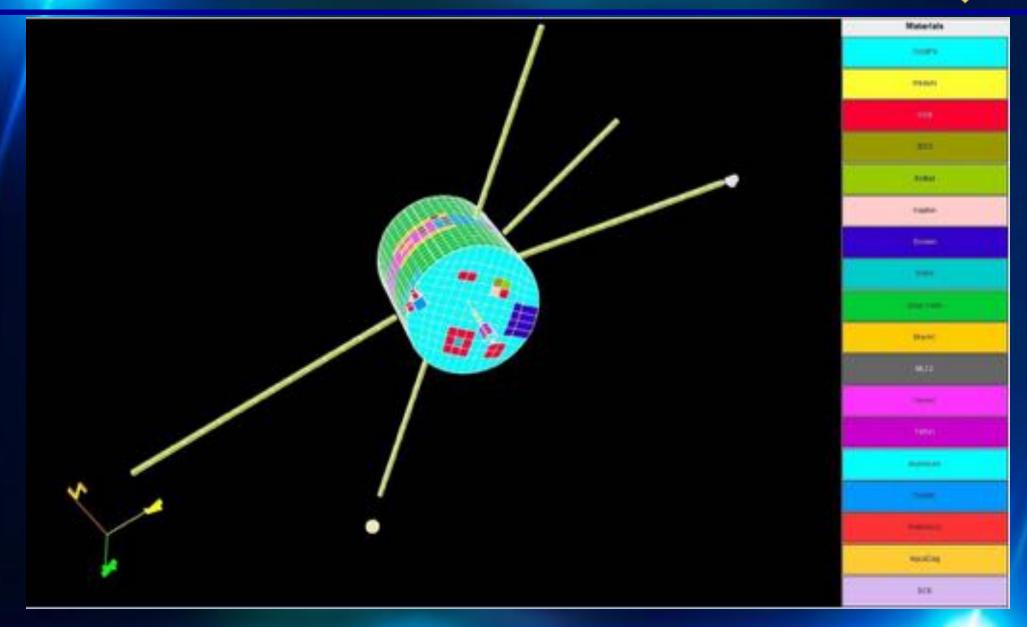
Geosynchronous Orbit Charging

in Script Edit Script	_		Prefix: Input File Name: InputFile Output File Name: OutputFile
Charge_Surfaces AppendChincl Coloutate_Matrices DefineInsulators DefineInsulators DoOneTimeSteps FutcolFoundFile ForConductorPotential Installator accutation Openitivativese PrepareChargeMatrix Pool_Matrices	>>Aut Correcamb-> Detate times Deplicate times	Surgit VShitt * C Charge_Burbanes x-scharps BEM_scharps x A smins x-scharps BEM_scharps x A smins x-scharps BEM_scharps x A prefit C Cpenclatic see • C Cpenclatic see mit albec arounds; • C SotThernend SotThernend • C SotThernend SotThernend • C SotThernend CotThersteps • C DoTimeSteps Filler	Current Command: DoTimeSteps Time: 0.319 Minimum Potential: -20.35 Maximum Potential: -11.91 Minimum Field: -103.4 Maximum Field: 12.54 Total Current: 6.995E-6 Elapsed Time: 141.4
ReadPhotocraission		Line Doost	

- Script (automatically generated) specifies calculation steps
 - Script can be edited internally or externally
 - Can do eclipse entry and exit, dynamic plasmas, spacecraft rotation, distributed array voltage
- Monitors progress of calculation
- Results in ~10 minutes on workstation





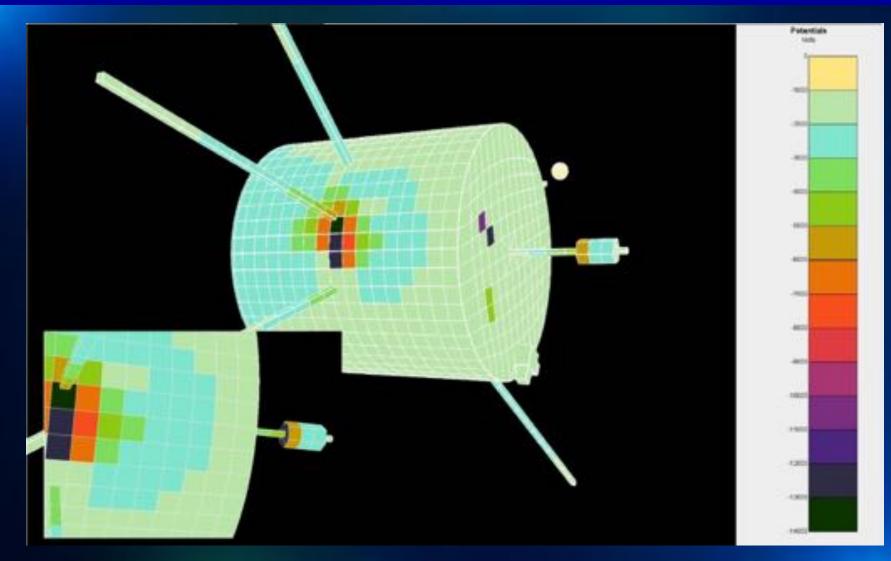


False color *Nascap-2k* surface-materials model of SCATHA Distribution A: Approved for public release; distribution unlimited



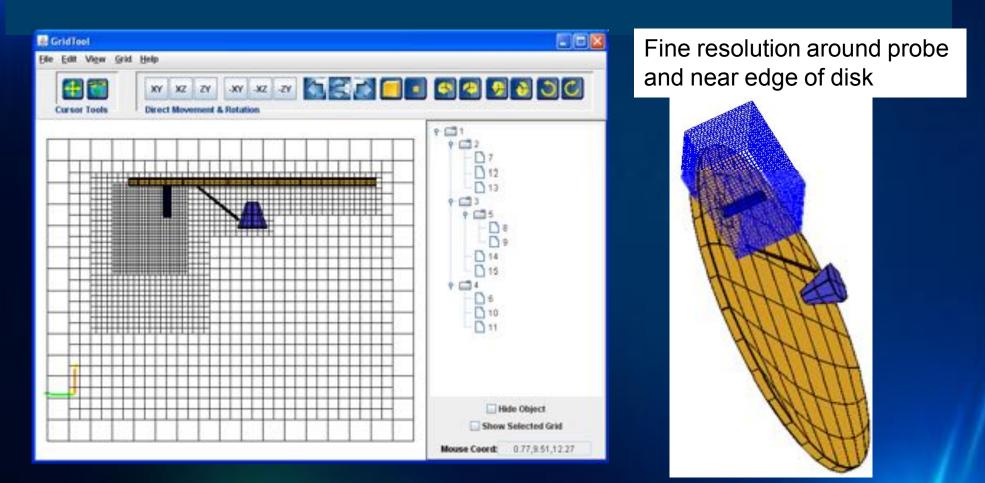
Charging of Spinning Spacecraft Nascap-2k Results for SCATHA





Potentials after 2010 seconds, ATS-6 environment, rotating at 1 rpm. Inset at lower left shows the bottomside of the Teflon cylindrical antenna cover, from which ions are partially blocked

Example - LEO Current Collection Spatial Gridding with *GridTool Charging Hazards and Wake Studies Experiment (CHAWS)*



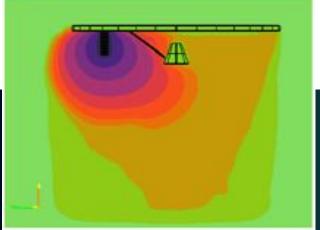
Multiply nested cubic grids with "special" elements containing object for computational speed

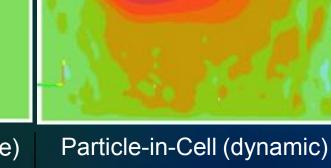


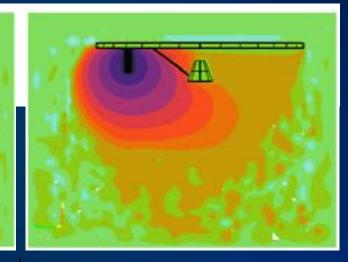
LEO Current Collection

Orbit Averaged Particle-In-Cell

 Macroparticle charge distributed over trajectory sub-steps, allows longer timesteps in dynamic calculations







Full Trajectory (steady-state) Macroparticles carry current Share *current* × *sub-step time* to grid *each sub-step*

10 iters (with sharing) in 2 hours

Macroparticles carry charge Share *charge* to grid at end of *timestep*

900 2-µs timesteps in 140 hours

Orbit averaged

Macroparticles carry charge Share $charge \times \frac{sub-step time}{timestep}$ to grid each sub-step 90 20-µs timesteps in 12 hours



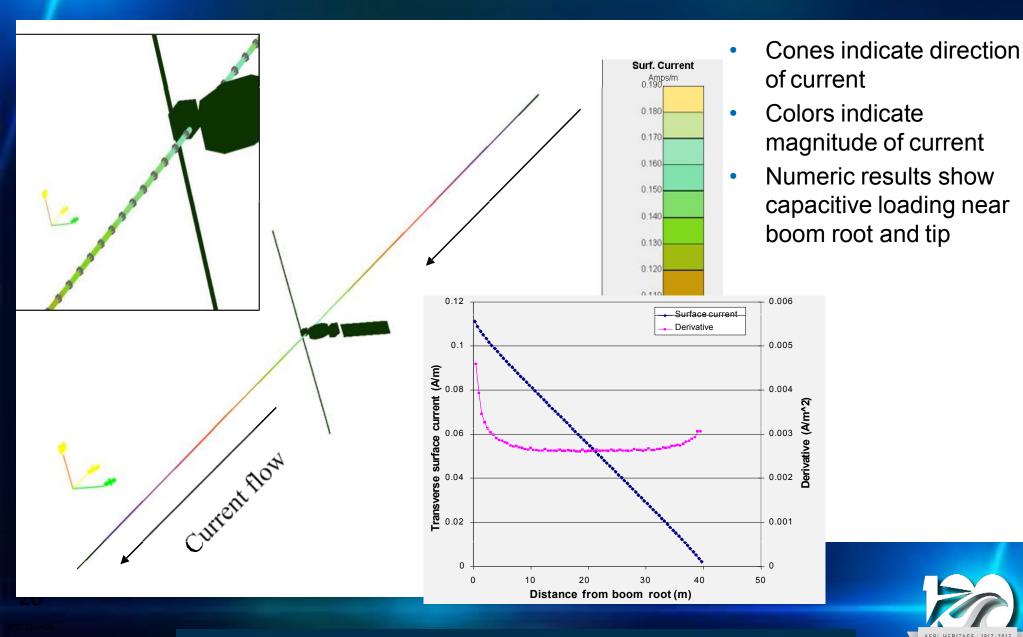
Potentials in Thruster Plumes

• Thruster plumes

- Produce potentials that modify contaminant trajectories
- Produce charge-exchange ions that lead to enhanced plasma density around the spacecraft
- Interact with spacecraft surfaces
- Interact with other thruster plumes
- Import plume ion densities from external file
 - Densities created by *PlumeTool*, part of EPIC (Electric Propulsion Interactions Code, a NASA SEE (Space Environments Effects) product)
- Calculate potentials self-consistently with charge exchange ion generation and transport



Antenna-induced Currents Surface Currents - DSX



Summary

- Nascap-2k
 - User-friendly integrated code
 - Study and analysis of a wide variety of spacecraft-plasma interactions
 - Variety of important space environments.
 - Uses efficient algorithms
 - Builds on heritage going back to late 1970s
- Examples presented
 - Charging in geostationary orbit
 - Current collection in low-Earth orbit
 - Charge exchange generation and potentials in thruster plumes
 - Surface and volume currents generated by antennae
- Nascap-2k is supported by Air Force Research Laboratory and the NASA Space Environments and Effects program
 - Distributed through http://see.msfc.nasa.gov

